Appendix A Requirements and Verification

Table 4: System Requirements and Verifications

Requirement	Verification	Point Allocation
1. Power Supply (a) Capable of handling 286A	1. Power Supply (a) Measure voltage difference	1. Power Supply
burst output* (b) Supply 12.6V ± 0.25V when fully charged	across power source using a Digital multimeter placed in parallel with the power supply. voltage should read $12.6V \pm 0.25V$	(a)/1
	(b) Measure current difference across power source and maximum load using a Digital	(b)/1
	multimeter placed in series with the power source and load that would draw 143A. The current should read 143A \pm 0.25A while maintaining the first tests temperature goals.	Total Points:/2
2. Transceiver (a) Capable of transmitting clean signal to receiver within 10 feet. Values $\pm 100 \mu S$ (b) Capable of actuating motors	 2. Transceiver (a) Turn on transmitter and check the RX values on the serial monitor. (b) Turn on transmitter and spin the motor using throttle's stick 	1. Transceiver (a)/2
		(b)/2
		Total Points:/4
3. Microcontroller (a) Capable of receiving 6 signals	3. Microcontroller (a) Display on Serial monitor that	1. Microcontroller
from the transmitter using a receiver (b) Capable of outputting PWM signals to the ESC	6 channels are being read (b) Hook up PWM outputs to oscilloscope and test if output signal looks right, and then	(a)/2
Signal to the Libe	connect this signal to an ES-C/motor to check that it correctly modulates the motors speed.	(b)/3
		Total Points:/5
		Continued on next page

Table 4 – continued from previous page

4. IMU module (accelerometer + gyro) (a) Capable of providing accurate angle measures in degrees with an accuracy of ± 10° (b) Capable of responding fast to changes of drone's position. Reading values at 100kHz ± 5kHz **Reading values at 100kHz ± 5kHz 5. Actuator (a) It must be able to be controlled by the transmitter and adjust an angle of 45°±5° 5. Actuator (a) It must be able to be controlled by the transmitter and adjust an angle of 45°±5° 6. Centrifugal Fans (a) Must generate a thrust of 455g (1.4 thrust/weight) ± 10g in conjunction with its motor running at 100% power. (b) Must weigh (housing + impeller)less than 100g ± 5g (c) Each of the centrifugal fans must weigh within ± 10 grams of each other 4. IMU module (accelerometer + gyro) (a) The test for angles consists of tilting one of the sensor's axis (e.g. roll) while maintaining the other axis fixed (e.g. pitch). Tilt 45° using a triangle to check the angle. Check the measurement obtained by the sensor on Arduino's Serial Monitor. (b) Defining in the code a 12C clock rate of 100kHz. Additionally, doing a simulation using the program called "Procesing" where is posible to stimulate how well responds the drone to changes in angle. 5. Actuator (a) Give the servo the signal for its 45° position, measure with precision angle measurement tool, then feed the servo the signal for its 45° position. Measure position again with the same tool 6. Centrifugal Fans (a) Must generate a thrust of 455g (1.4 thrust/weight) ± 10g in conjunction with its motor running at 100% power. (b) Must weigh (housing + impeller)less than 100g ± 5g (c) Each of the centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of each other centrifugal fans must weigh within ± 10 grams of	Requirement	Verification	Point Allocation
5. Actuator (a) It must be able to be controlled by the transmitter and adjust an angle of $45^{\circ}\pm5^{\circ}$ (a) Give the servo the signal for its 45° position, measure with precision angle measurement tool, then feed the servo the signal for its -45° position. Measure position again with the same tool 6. Centrifugal Fans (a) Must generate a thrust of $455g$ (1.4 thrust/weight) $\pm 10g$ in conjunction with its motor running at 100% power. (b) Must weigh (housing + impeller)less than $100g \pm 5g$ (c) Each of the centrifugal fans must weigh within ± 10 grams of each other (a) Give the servo the signal for its 45° position, measure with precision angle measurement tool, then feed the servo the signal for its 45° position. Measure position again with the same tool 6. Centrifugal Fan (a) Attach motors to centrifugal Fan (a) Attach motors to centrifugal Fan (b) Memodula compressors and determine thrust by measuring force (in Newtons) of each individual compressor thrusting itself against the pad. (c) Each of the centrifugal fans must weigh within ± 10 grams of each other (b) Weigh module on a scale to determine mass measurements between each modula	 4. IMU module (accelerometer + gyro) (a) Capable of providing accurate angle measures in degrees with an accuracy of ± 10° (b) Capable of responding fast to changes of drone's position. Reading values at 100kHz ± 	 4. IMU module (accelerometer + gyro) (a) The test for angles consists of tilting one of the sensor's axis (e.g. roll) while maintaining the other axis fixed (e.g. pitch). Tilt 45° using a triangle to check the angle. Check the measurement obtained by the sensor on Arduino's Serial Monitor. (b) Defining in the code a I2C clock rate of 100kHz. Additionaly, doing a simulation using the program called "Procesing" where is posible to simulate how well responds the 	1. IMU module (a)/4 (b)/3
(a) Must generate a thrust of 455g (1.4 thrust/weight) ± 10g in conjunction with its motor running at 100% power. (b) Must weigh (housing + impeller)less than 100g ± 5g (c) Each of the centrifugal fans must weigh within ± 10 grams of each other (a) Attach motors to centrifugal gal compressors and determine thrust by measuring force (in Newtons) of each individual compressor-motor pair on a stationary force sensor pad with the compressor thrusting itself against the pad. (b) Weigh module on a scale to determine mass measurements between each module	(a) It must be able to be controlled by the transmitter and adjust	5. Actuator (a) Give the servo the signal for its 45° position, measure with precision angle measurement tool, then feed the servo the signal for its -45° position. Measure position again with	(a)/2
Continued on next page	 (a) Must generate a thrust of 455g (1.4 thrust/weight) ± 10g in conjunction with its motor running at 100% power. (b) Must weigh (housing + impeller)less than 100g ± 5g (c) Each of the centrifugal fans must weigh within ± 10 grams 	 (a) Attach motors to centrifugal compressors and determine thrust by measuring force (in Newtons) of each individual compressor-motor pair on a stationary force sensor pad with the compressor thrusting itself against the pad. (b) Weigh module on a scale to determine mass meets requirements (c) Compare mass measurements 	(a)/7 (b)/2 (c)/1 Total Points:/4

Table 4 – continued from previous page

	- continued from previous page	D
Requirement	Verification	Point Allocation
 7. Flight Controller (a) Maintaining 0 ±5 setpoint (roll, pitch leveled) and changing sensor values, PID outputs must change (b) Control must update values at 100Hz ± 20Hz (sensor speed) (c) A good tunning of PID parameters must maintain the drone leveled parallel to the floor at 0°± 10° 	 7. Electronic Speed Controller (a) Check in the Serial Monitor that PID outputs changes depending on the sensor readings (b) Check in the Serial Monitor that PID outputs are updating at the same speed that sensors do (c) Make the drone fly leveled after pitching or rolling 	1. Flight Controller (a)/2 (b)/2 (c)/4
8. Motors	8. Motors	Total Points:/8 1. Motors
(a) Capable of providing 455g±10g of overall lift in junction with centrifugal fan	(a) Attach motors to centrifugal compressors and determine thrust by measuring force (in grams) of each individual compressor-motor pair on a stationary force sensor pad with the compressor thrusting itself against the pad.	(a)/5 Total Points:/5
9. Chassis Frame (a) Capable of withstanding a fall of 1m ± 0.1m (b) Must weigh in overall (including electronics and fans) less	9. Chassis Frame (a) Drop drone from 1.1m and repeat at 0.9m (b) Weigh drone on a scale. It must weigh less than	1. Chassis Frame (a)/1
ing electronics and rans) less that $1.4 \text{kg} \pm 0.1 \text{kg}$	1.4kg ± 0.1 kg	(b)/3
		Total Points:/8
		Continued on next page

Table 4 – continued from previous page

Requirement	Verification	Point Allocation
9. PCB (a) Output signals biased from $0\pm0.25\mathrm{V}$ to $3\pm0.3\mathrm{V}^*$ (b) Voltage alarm rings and lights up LED when the input voltage is less than $10.6\pm0.25\mathrm{V}$	9. PCB (a) Hook up power supply and use multimeter to measure voltage at output pins (b) SUpply the PCB with 10.8V with power source and see if LED lights up and alarm rings	1. PCB (a)/1 (b)/2
		Total Points:/3
		Total Points/50

^{*3.}a The six signal corresponds to the six commands that you have in our transceiver.

^{*9.}a The Microcontroller may only take a maximum voltage input of $3.3\mathrm{V}$ to prevent damage to the board.

Appendix B Circuit Schematic

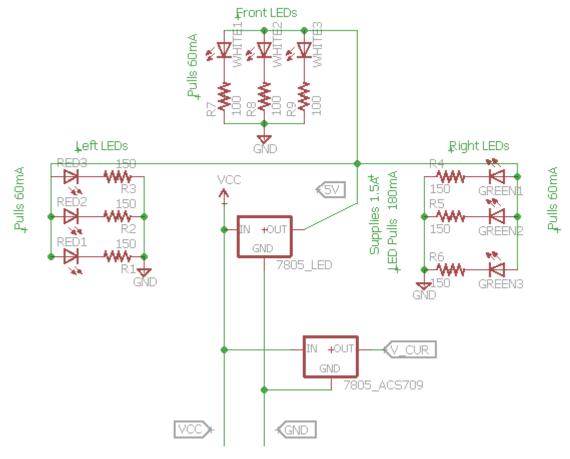


Figure 13: LED module to display orientation of drone

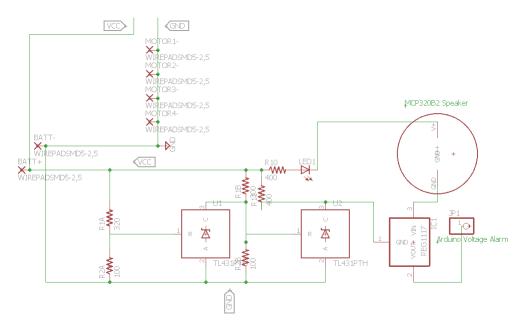


Figure 14: Voltage Alarm module and Mounting pads for motors

Hall Effect Current Sensor ACS709 U6 MOTORI+IP1+ NC4 WIREPADSMD5-2,5 NC3 23B 22B IP3+ FAULT_EN 21B IP4+ VOC 20B IP5+ VCC 19B IP6+ FAULT 18B IP1-VIOUT 17B FILTER IP2-VCC 16B VZCR IP3-15B GND 14B IP5-NC2 13B NC1 мотвз гы+ NC4

Figure 15: One of the four ACS709 Chips used for sensing current $\,$

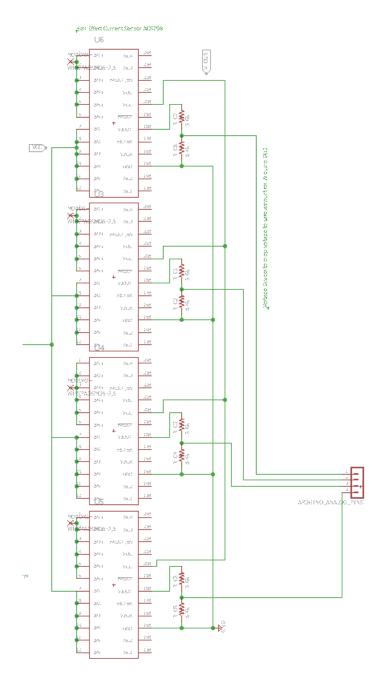


Figure 16: All four of the ACS709 Chips used for sensing current

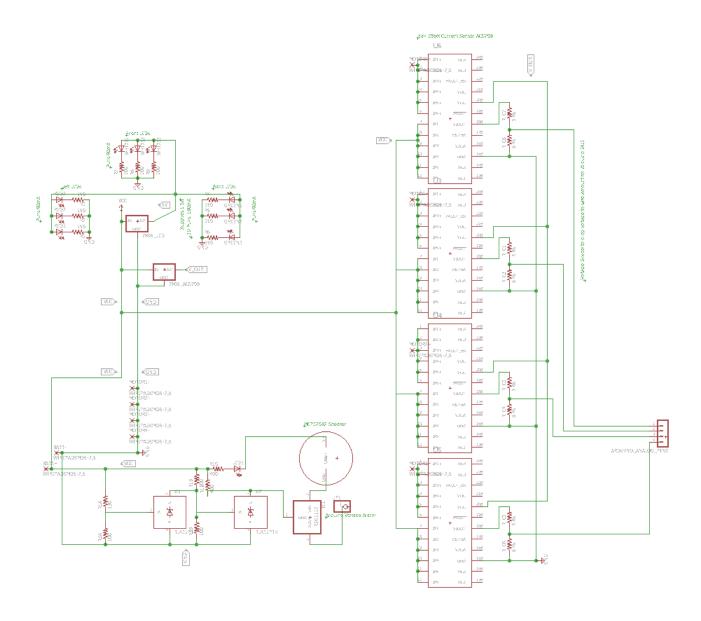


Figure 17: Complete Circuit

B.1 PCB Design

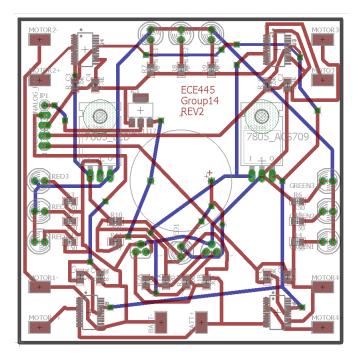


Figure 18: PCB design from EAGLE CAD

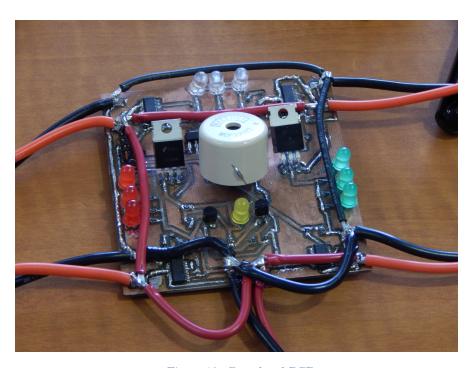


Figure 19: Completed PCB

Appendix C Software Flowchart

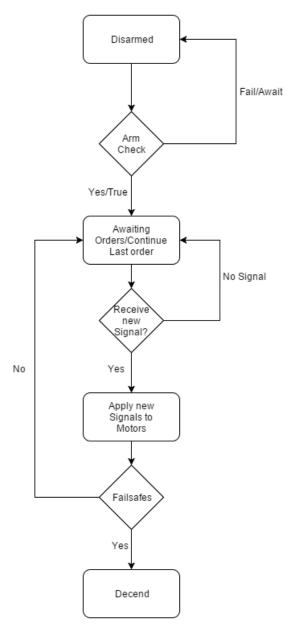


Figure 20: Flowchart for drone

Appendix D Schedule

Table 5: Schedule

Week	Task Responsibility	
9-Feb	 Finalize Proposal Study dynamics of drone Prepare mock design review 	 Leo and Bree Bree and Ignacio Everybody
15-Feb	CAD Centrifugal fan Design ESC Prepare design review	 Ignacio and Leo Bree Everybody
22-Feb	 Purchase microcontroller Purchase sensors and test Purchases motors and test Purchase centrifugal fan materials and prototype Finalize Design Review 	 Leo Bree Ignacio Leo Everybody
29-Feb	 Assemble Control Mechanism Reiterate centrifugal design Assembly prototype propulsion mechanism Assemble acutuation and Chassis 	 Leo Leo Ignacio Bree
7-Mar	 Prepare mock demo Finalize PCB Design Run more tests on control polish centrifugal fan design 	 Ignacio Bree Leo Leo
14- Mar	 Finalize mock demo Polish drone design Run final test on drone capability Prepare individual progress report 	 Everybody Bree and Ignacio Leo Everybody
4-Apr	 Fix any remaining issues Prepare presentation Prepare demonstration 	1. Leo 2. Bree 3. Ignacio
		Continued on next page

Table 5 – continued from previous page

Week	Task	Responsibility
11-		
Apr	1. Prepare Final paper	1. Everybody
	2. Finalize presentation	2. Bree
	3. Finalize demonstration	3. Ignacio
2-May	 Finalize Final paper Lab checkout 	1. Everybody 2. Leo
2-May	1	

Appendix E Testing Apparatus

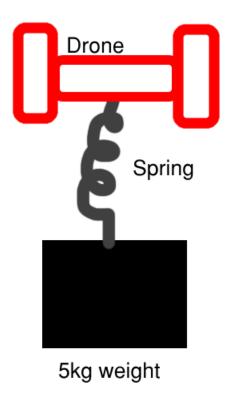


Figure 21: Testing apparatus for on-campus testing

Appendix F Code Snippets

```
typedef struct control_state{
       //holds the updated bit flags
       volatile unsigned char bUpdateFlagsShared;
       volatile unsigned short unThrottleInShared;
       volatile unsigned short unRudderInShared;
       volatile unsigned short unElevatorInShared;
       volatile unsigned short unAileronInShared;
       volatile unsigned short unGearInShared;
       unsigned char bUpdateFlags;
       unsigned int ulThrottleStart=0;
       unsigned int ulGearStart=0;
       unsigned int ulRudderStart=0;
       unsigned int ulElevatorStart=0;
       unsigned int ulAileronStart=0;
       int throttle;
       int rudder;
       int elevator;
       int aileron;
       int gear;
       //Boolean to see if drone is amred
       bool armed = false;
       //Boolean to see if signal is good
       bool signalGood = false;
       //PID values
       double pid_roll_in, pid_roll_out, pid_roll_setpoint = 0;
       double pid_pitch_in, pid_pitch_out, pid_pitch_setpoint = 0;
       double pid_yaw_in, pid_yaw_out, pid_yaw_setpoint = 0;
       //Angles values for the sensor
       double angleX = 0;
       double angleY = 0;
       double angleZ = 0;
}control_state_t;
```

Figure 22: Control State Code

```
//Function to read in the signals
uint8 t COMMS::readSignals(control state *ctrl)
  if (ctrl->bUpdateFlagsShared)
    //turn off interrupts to take local copies of the sharead variables
   noInterrupts();
    //take local copy of which channels were updated
    ctrl->bUpdateFlags = ctrl->bUpdateFlagsShared;
    if(ctrl->bUpdateFlags & THROTTLE_FLAG)
      ctrl->throttle = ctrl->unThrottleInShared;
    if(ctrl->bUpdateFlags & AILERON_FLAG)
      ctrl->aileron = ctrl->unAileronInShared;
    if(ctrl->bUpdateFlags & ELEVATOR_FLAG)
      ctrl->elevator = ctrl->unElevatorInShared;
    if(ctrl->bUpdateFlags & RUDDER_FLAG)
      ctrl->rudder = ctrl->unRudderInShared;
    ctrl->bUpdateFlagsShared = 0;
    interrupts();//Turn interrupts back on
 ctrl->bUpdateFlags = 0;
```

Figure 23: Communication Code - Read Signals

```
//Function to calculate the value read by the receiver
void calcThrottle()
{
   if (digitalRead(THROTTLE_PIN) == HIGH)
   {
      my_state.ulThrottleStart = micros();
   }
   else
   {
      my_state.unThrottleInShared = (short)(micros() - my_state.ulThrottleStart);
      my_state.bUpdateFlagsShared |= THROTTLE_FLAG;
   }
}
```

Figure 24: Communication Code - Calculate Signal

```
//Set motor velocities in terms of miliseconds
uint8_t MOTORS::updateMotors(int front, int back, int right, int left)
{
    /*
    * m0 = front left = throttle + pid_roll_out
    * m1 = front right= throttle + pitch
    * m2 = back right = throttle - pid_roll
    * m3 = back left = throttle - pid_pitch
    */
    ESC1.writeMicroseconds(front);
    ESC2.writeMicroseconds(back);
    ESC3.writeMicroseconds(right);
    ESC4.writeMicroseconds(left);
}
```

Figure 25: Motor Control Code

```
//Function to update the motor values
void updateCont()
  int m0, m1, m2, m3;
  int throttle value;
  setpoint update();
 pid_update();
 pid compute();
  //Mapping values to useable range
  throttle value = map (my state.throttle, SIGNAL MIN, SIGNAL MAX, MOTOR MIN, MOTOR MAX);
  m0 = throttle_value + my_state.pid_roll_out;
 ml = throttle value + my state.pid pitch out;
  m2 = throttle_value-50 - my_state.pid_roll_out;
  m3 = throttle_value - my_state.pid_pitch_out;
  my motors.updateMotors(m0, m1, m2, m3);
  //Serial.println(m0);
  my_motors.updateServos(my_state.rudder);
  }
```

Figure 26: Updating motors with controller

Appendix G CAD Models

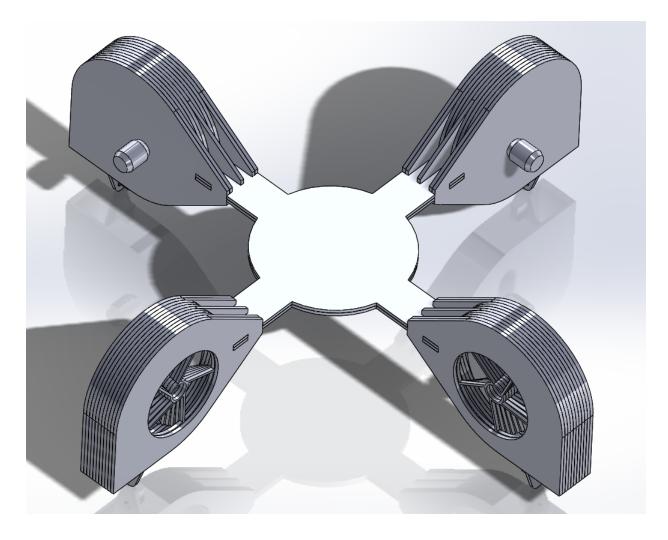


Figure 27: Isometric view of Drone CAD

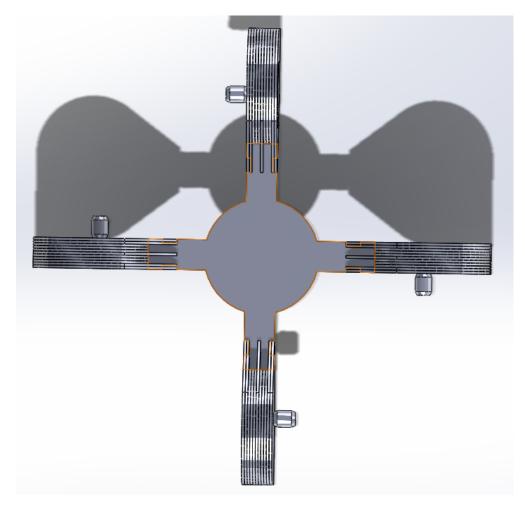


Figure 28: Top View of Drone CAD

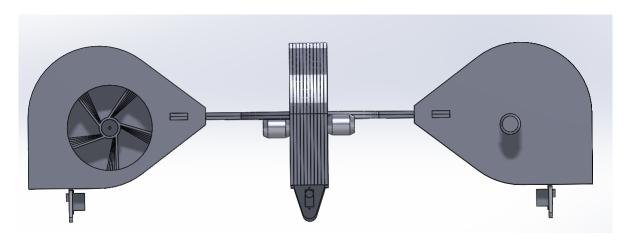


Figure 29: Side View of Drone CAD

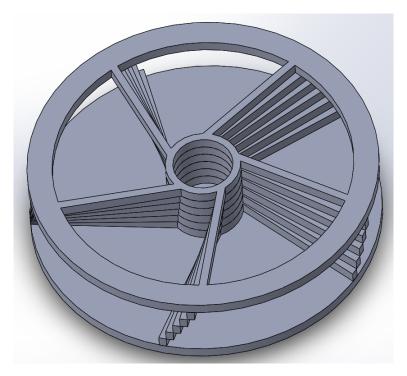


Figure 30: Impeller for Counter Clockwise Fan

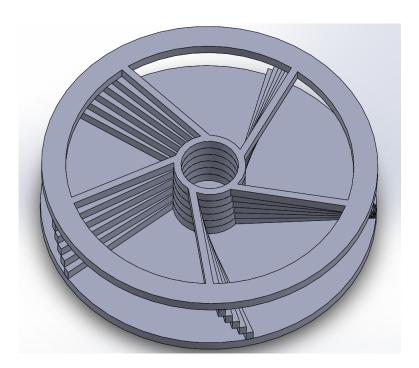


Figure 31: Impeller for Clockwise Fan



Figure 32: Turbine Housing for Counter Clockwise Fan



Figure 33: Turbine Housing for Clockwise Fan

Appendix H Sources

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